

TEST CASE DOCUMENTATION AND TESTING RESULTS

LSTC-QA-LS-DYNA-AEROQA-10-4

TEST CASE ID AEROQA-10

MAT_224

Dynamic Punch Test Aluminium 2024

Tested with LS-DYNA® R6.1.0 Revision 74904

Wednesday 5th September, 2012

Warranty Disclaimer:

The test case(s) described herein are for illustrative purposes only. LSTC does not warrant that a user of these or other LS-DYNA features will experience the same or similar results or that a feature will meet the user's particular requirements or operate error free. FURTHERMORE, THERE ARE NO WARRANTIES, EITHER EXPRESS OR IMPLIED, ORAL OR WRITTEN, WITH RESPECT TO THE DOCUMENTATION AND SOFTWARE DESCRIBED HEREIN INCLUDING, BUT NOT LIMITED TO ANY IMPLIED WARRANTIES (i) OF MERCHANTABILITY, OR (ii) FITNESS FOR A PARTICULAR PURPOSES, OR (iii) ARISING FROM COURSE OF PERFORMANCE OR DEALING, OR FROM USAGE OF TRADE OR. THE REMEDIES SET FORTH HEREIN ARE EXCLUSIVE AND IN LIEU OF ALL OTHER REMEDIES FOR BREACH OF WARRANTY.

Document Information	
Confidentiality	external use
Document Identifier	LSTC-QA-LS-DYNA-AEROQA-10-4
Author(s)	Prepared by LS-DYNA® Aerospace Working Group
Number of pages	12
Date created	Wednesday 5 th September, 2012
Distribution	LS-DYNA® Aerospace Working Group / internal LSTC QA

Contents

1	Introduction	1
1.1	Purpose of this Document	1
2	Test Case Information	2
3	Test Case Specification	3
3.1	Test Case Purpose	3
3.2	Test Case Description	4
3.3	Model Description	5
4	Test Specifications	7
4.1	Test Case Targets	7
4.2	Pass/Fail Criteria	7
5	Test Case Results	8
5.1	Software and Hardware Specifications	8
5.2	Results Summary	9
5.3	Result Details	10
5.3.1	Sub Test Case ID 1 - Test Target 1	11
5.3.2	Sub Test Case ID 2 - Test Target 1	11
	References	12

1 Introduction

1.1 Purpose of this Document

This document specifies the test case AEROQA-10. It provides general test case information like name and ID as well as information to the confidentiality, status, and classification of the test case.

A detailed description of the test case is given, the purpose of the test case is defined, and the tested features are named. The test case specifications also state the target measures for testing and the expected results, as well as their pass and fail criteria.

Testing results are provided in section 5 for the therein mentioned LS-DYNA® version and platforms.

2 Test Case Information

Test Case Summary	
Confidentiality	external use
Test Case Name	MAT_224 Dynamic Punch Test Aluminium 2024
Test Case ID	AEROQA-10
Test Case Status	active
Test Case Classification	Example
Test Case Source	NCAC/GWU
Tested Keyword	*MAT_TABULATED_JOHNSON_COOK
Member of Test Suite	AEROQA SUITE
Metadata	AEROQA

Table 1: Test Case Summary

3 Test Case Specification

3.1 Test Case Purpose

The purpose of QA Test Case ID AEROQA-10 is the comparison of results from different cpu architectures for punch tests of Aluminium 2024.

The reliability and consistency of LS-DYNA® as a finite element solver for this punch test simulation is evaluated by performing analyses on different cpu architecture platforms.

3.2 Test Case Description

This QA Test Case contains punch tests (see figure 1) performed on a Split Hopkinson Bar (SHB) which are used to examine the failure of Aluminium 2024.

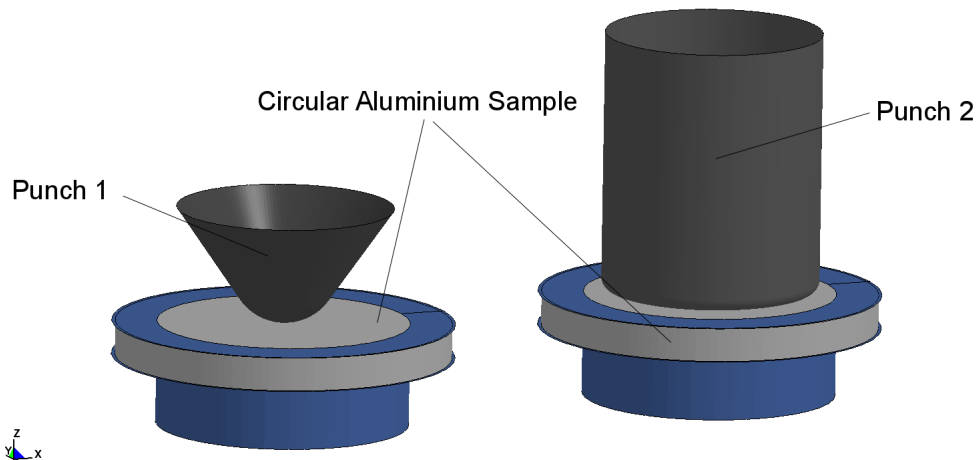


Figure 1: Model sketch: Punch test on circular Aluminium samples with two different punch shapes

Table 2 contains a short summary of the physical model set up.

Physical Model Information	
circular sample geometry	diameter = 14.56 mm, thickness = 1.456 mm
sample material	Aluminium 2024
punch velocity	20 m/sec

Table 2: Model set-up data

3.3 Model Description

The model geometry is discretized with solid elements for the circular Aluminium 2024 sample and shell elements for the punch geometry (see figure 2).

The model specifications can be found in table 3, and table 4 defines the sub test case specification.

The material definitions and their parameters can be found in the input decks.

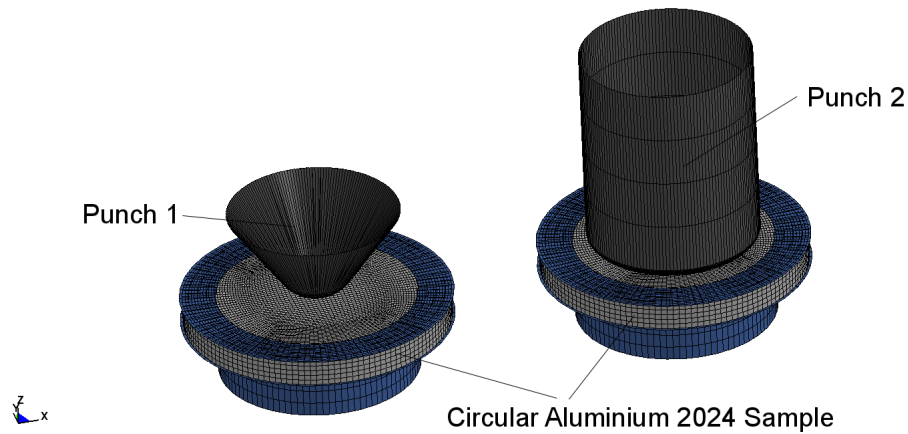


Figure 2: FEA model: Punch test on circular Aluminium samples with two different punch shapes

FEA Model information		
Sub Test Case ID ¹	1	2
Nodes	32882	34544
Solid elements	23625	23625
Solid materials	1	1
Shell elements	3910	5598
Shell materials	3	3
Parts	4	4
Units	mm (length), s (time), tonne (mass), N/mm ² (stress), Nmm (energy)	

¹ Sub Test Case ID refers to the ID's in table 4

Table 3: FEA Model Information

Sub Test ID	Punch Type	Input Deck Name
1	Punch 1	pch1_mod.k
2	Punch 2	pch6_mod.k

Table 4: Specification of sub test cases

4 Test Specifications

4.1 Test Case Targets

Table 5 displays the test case targets. The test case targets specify values or a series of values taken from the finite element analysis solution of the test case and they are used in a comparison of analysis results on different cpu architectures. They are chosen in a way that they are representative of the numerical model.

Test Case Targets				
Target number	output	component type	component id	retrieved from
1	resultant interface force	z	2	binout/rcforc file

Table 5: Test Case targets for QA Test Case ID AEROQA-10

Test case targets are used to evaluate the cross cpu architecture consistency (see section 4.2).

4.2 Pass/Fail Criteria

These are the Pass/Fail criteria used for the cross cpu architecture consistency test of the Test Case ID AEROQA-10.

The sub test case passes if all of the following criteria are reached:

- For a specific test case target, the maximum distance between an x-y pair of a slope of one cpu architecture/software version combination to at least one x-y pair of all other tested cpu architecture/software version combinations is 15% of the maximum slope value.

Otherwise the cross cpu architecture consistency test fails.

5 Test Case Results

5.1 Software and Hardware Specifications

In order to ensure cross-platform consistency, the herein mentioned sub test cases are run on platforms specified in table 6 and the results are calculated with software versions defined in table 7.

Platform Name	Operating system	CPU type	MPI-Protocol	Number of cpu's ¹
sandwich	SUSE LES 11.1	Intel® Xeon® E7- 8837 @ 2.67GHz	hpmapi	4
ham	CentOS 5.4	AMD® Opteron® 8435 @ 800MHz	hpmapi	4
sgi64d	SUSE LES 9.4	Intel® Itanium® 2 @ 1.6GHz	hpmapi	4

¹ Number of cpu's used for calculation of the test case

Table 6: Used Platforms and CPU Type's

Product	Version	Release	Revision	Parallel type ¹	Precision ²	executable
LS-DYNA®	971	R6.1.0	74904	SMP	SP	ls971.74904.R6.1.0
LS-DYNA®	971	R6.1.0	74904	SMP	DP	ld971.74904.R6.1.0
LS-DYNA®	971	R6.1.0	74904	MPP	SP	mpp971.74904.R6.1.0
LS-DYNA®	971	R6.1.0	74904	MPP	DP	mpd971.74904.R6.1.0

¹ MPP = Massively Parallel Processing, SMP = Symmetric Multiprocessing

² SP = single precision, DP = double precision

Table 7: Tested LS-DYNA® version

5.2 Results Summary

Table 8 contains the results of the QA Test Case ID AEROQA-10 completed with all combinations of software and hardware defined in section 5.1 (2 * 3 * 4 total calculation runs).

Details on the test results can be found in the section 5.3.

The table 8 cross cpu architecture consistency summary is:

- **PASS** - Pass criteria in section 4.2 is attained.
- **FAILED** - Pass criteria in section 4.2 is not attained.
- **ERROR** - sub test case terminates due to error.
- **N/A** - sub test case was not calculated.

Sub Test Case ID	PASS/FAILED
1	PASS
2	PASS

Table 8: Results summary for QA Test Case ID AEROQA-10

5.3 Result Details

The following subsections contain detailed results for the QA Test Case ID AEROQA-10 for LS-DYNA® R6.1.0 Revision 74904.

For each sub test case defined in section 3.3 there is a graph displaying the time history of the result target defined in section 4.1 for the platform and software version combinations defined in section 5.1.

The title of the graph states the name of the input deck, the result file name, and the output separated by underscores. The legend contains the result file name, output, platform, and executable. The last number of the legend specifies the number of cpu's used to calculate the example. A leading minus sign refers to the compatibility option for SMP calculations (see [1] for details on this option).

Example for title and legend:

Title:

'pch1_mod.k: rforc_slave_x_2' states that the input deck for sub test case 1 was used to calculate these results. The component displayed is the resultant force in x-direction derived from the 'rforc' output file.

Legend:

'rforc_slave_x_2_sandwich_ls971.74904.R6.1.0_4' states that the graph shows the resultant force in x-direction derived from the 'rforc' output file for an input deck which was calculated on the 'sandwich' platform with a LS-DYNA® R6.1.0 Revision 74904 binary (SMP, single precision) on four processors.

5.3.1 Sub Test Case ID 1 - Test Target 1

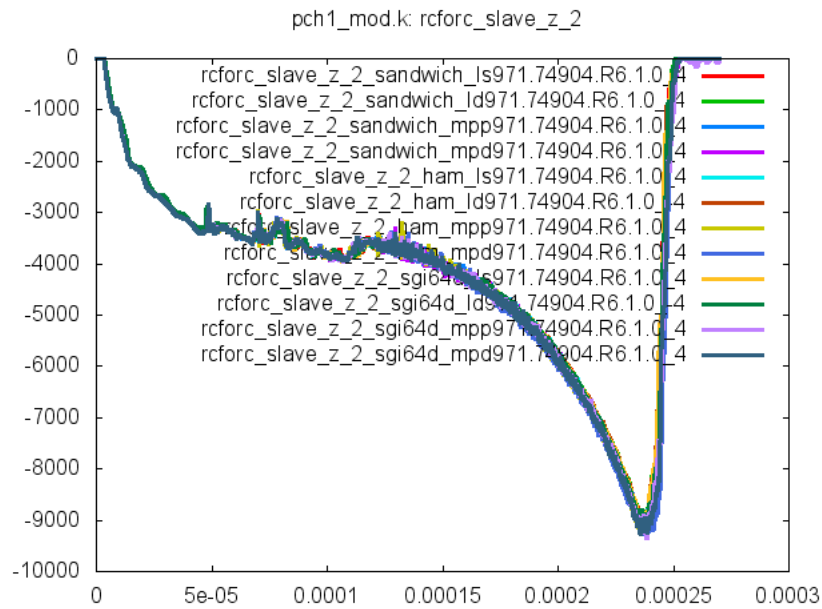


Figure 3: Cross platform results, resultant force in z-direction, sub test case ID 1

5.3.2 Sub Test Case ID 2 - Test Target 1

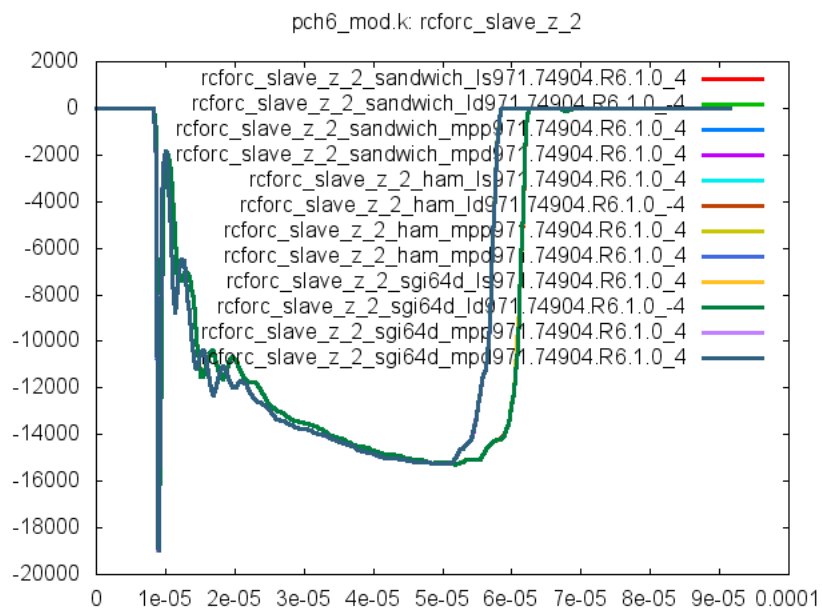


Figure 4: Cross platform results, resultant force in z-direction, sub test case ID 2

References

- [1] LSTC, *LS-DYNA KEYWORD USER MANUAL*, 7374 Las Positas Road, Livermore, CA, 94551, USA, version 971 ed., May 2007.